TECHNICAL REQUIREMENTS FOR HELICOPTER LANDING GRID

1. **Purpose**. Landing grid is used to secure the Helicopter against flight deck.

- Design.
 - (a) Design Standard NATO's STANAG 1276
 - (b) Landing Grid is to be designed for the following parameters: -
 - (i) Traction Weight 11 MT
 - (ii) Permissible Compression load 21500 Kgs
 - (iii) Overall Diameter 2750mm.
 - (iv) Useful Diameter 2500mm.
 - (v) Total maximum weight of Grid 3000 Kgs.
 - (vi) Depth should not exceed 200mm.
 - (c) The Landing Grid is to be supplied with suitable drain arrangement, Rubber/ Neoprene cover & securing arrangement.
 - 3. <u>Environmental Conditions</u>. The equipment should function smoothly under extreme tropical conditions as mentioned below: -

Ambient Air temperature 45 Degrees C Sea Water temperature 35 Degrees C

Relative humidity 100% at 35 Degrees C

Roll +/- 30 Degrees
Roll Period 10 Seconds
List +/- 20 Degrees
Pitch +/-7 Degrees
Pitching period 10 Seconds.

- 4. **Shock Test**. The equipment and its accessories should be capable of withstanding shock accelerations as per Shock Grade NSS-II. The actual shock calculations or shock test results should be furnished as a support of proof.
- 5. <u>VibrationTest</u>. The parts of the equipment and its accessories are to be designed to ensure resistance to fracture, distortion or misalignment due to forces of vibration.
- 6. <u>Material Specification</u>. The material of construction of Grid shall be Stainless Steel. The properties are as detailed below: -
 - (a) Mechanical Properties.

Yield stress - 8000 Kgf/cm2 UTS - 9500 Kgf/cm2

Impact strength - 35 Joules/cm2 (20 deg C)

Elongation - 14% on 5D

(b) **Chemical Composition**.

 Carbon
 - 0.06% (max)

 Manganese
 -1.0% (max)

 Silicon
 -0.4% (max)

 Nickel
 -4.5-6.0 %

 Chromium
 -14.5-17.0 %

 Molybdenum
 -0 .7-1.1 %

7. <u>Tests and Trials</u>. Test load is to be applied at the Centre of grid. The same has to be calculated considering Harpoon traction of 11 tons and compression load of 21.5 tons.